

Effects of Moisture Diffusion in Sandwich Composites

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UW Center for Urban Horticulture

Effects of Moisture Diffusion in Sandwich Composites

Motivation and Key Issues:

- In-service bond failures between composite facesheets and honeycomb cores have been reported

X-33 Liquid Hydrogen
Tank Failure

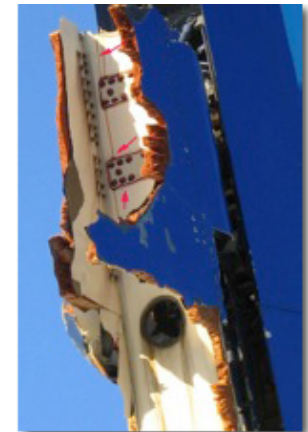


Boeing 747 upper
skin disbonds



approx. 24" x 60"
upper skin disbond

Airbus A-310
Rudder Failure



(Photos courtesy of Ronald Krueger, National Institute of Aerospace)

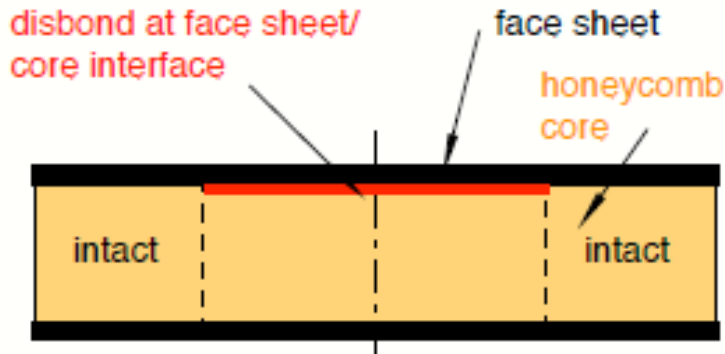
Effects of Moisture Diffusion in Sandwich Composites

Motivation and Key Issues:

- Disbond initiation and growth is not completely understood, but is thought to occur due to combination of factors:
 - Pressure differences between inside and outside of unvented honeycomb structures (Ground-Air-Ground or 'GAG' pressure cycles)

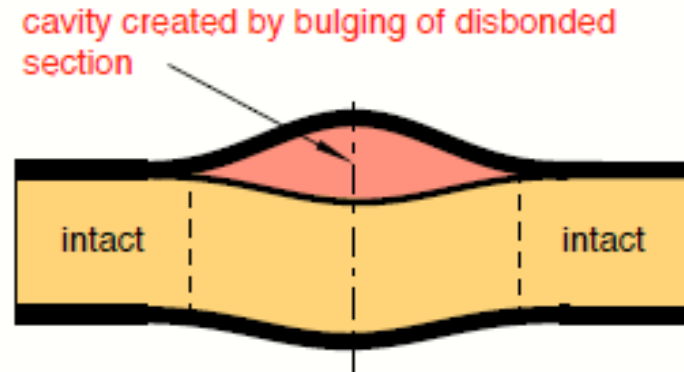
Configuration at ground level

$$P_o = 100 \text{ kPa} = 14.7 \text{ psi}$$



Configuration at 35,000 ft

$$P_o = 24 \text{ kPa} = 3.5 \text{ psi}$$



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- Water ingress most commonly attributed to wicking of liquidous water through microcracks, along fiber/matrix interface, and/or through improper edge closeouts (all accentuated by GAG pressure cycles)
- Water ingress may also occur simply due to diffusion of water molecules through (undamaged) facesheets

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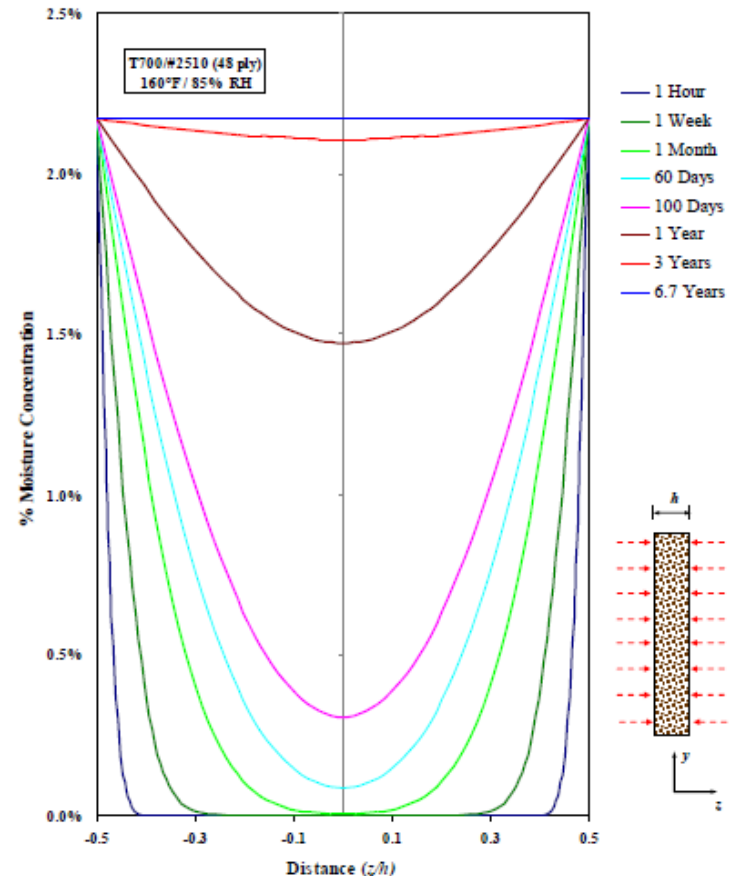
Motivation and Key Issues:

- Significant moisture transport via diffusion typically requires months or years, depending on:
 - Temperature
 - Thickness and material properties
 - External humidity level
-

Effects of Moisture Diffusion in Sandwich Composites

Motivation and Key Issues:

Moisture diffusion in solid 48-ply Gr-Ep laminate; 160°F, 85%RH
(W. Seneviratne and J. Tomblin, JAMS 2012)



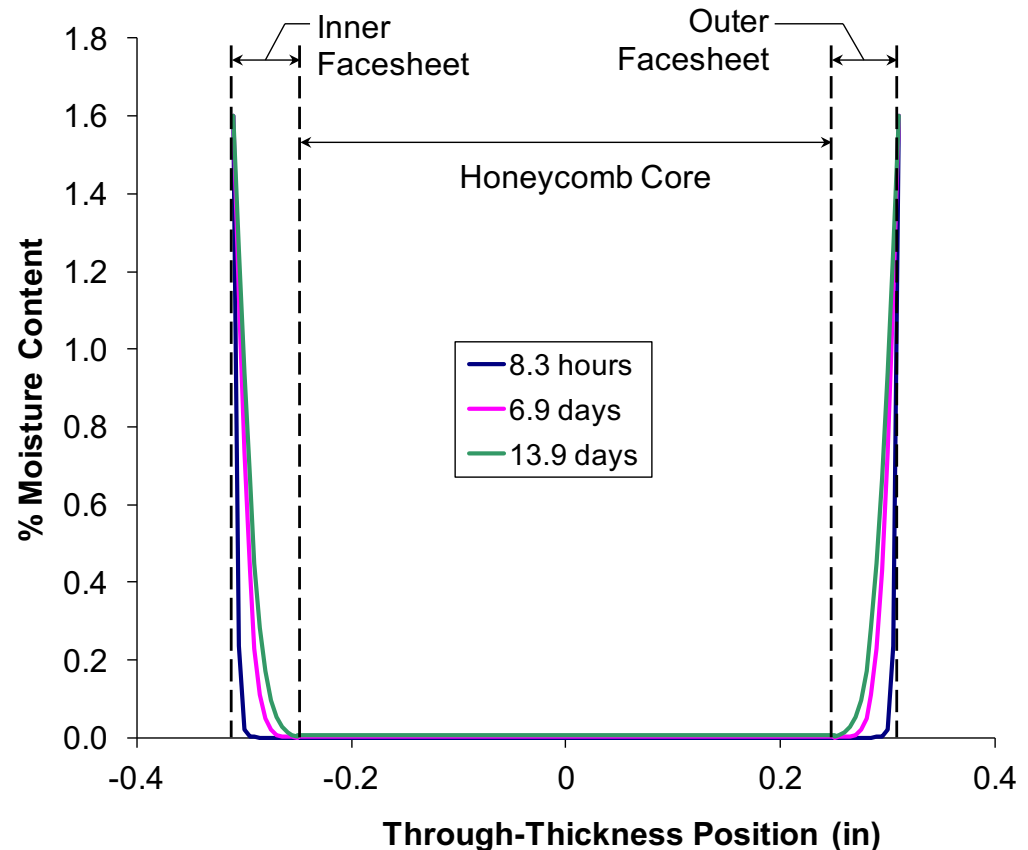
Effects of Moisture Diffusion in Sandwich Composites

Motivation and Key Issues:

Moisture diffusion in honeycomb sandwich panels:

- 12-ply Gr-Ep facesheets
- 0.5 in Nomex core
- 90°F, 80%RH

(Tuttle, AMTAS 2009)



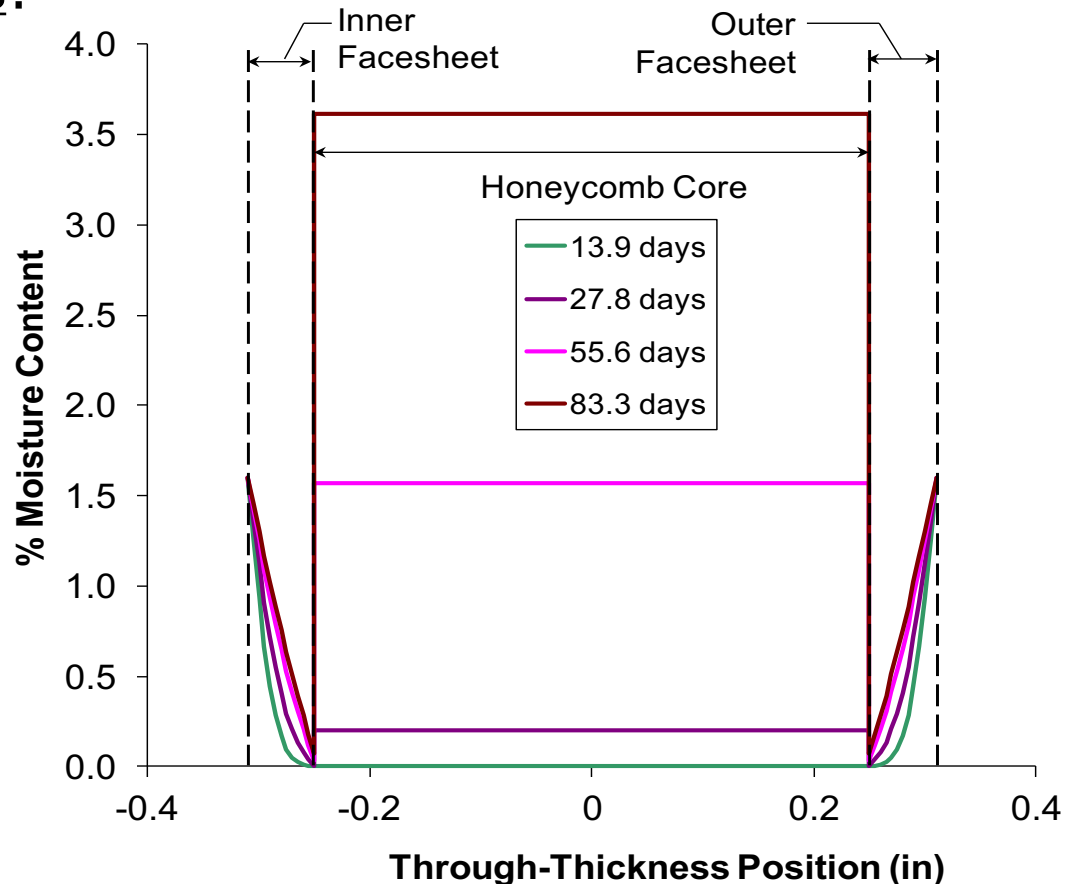
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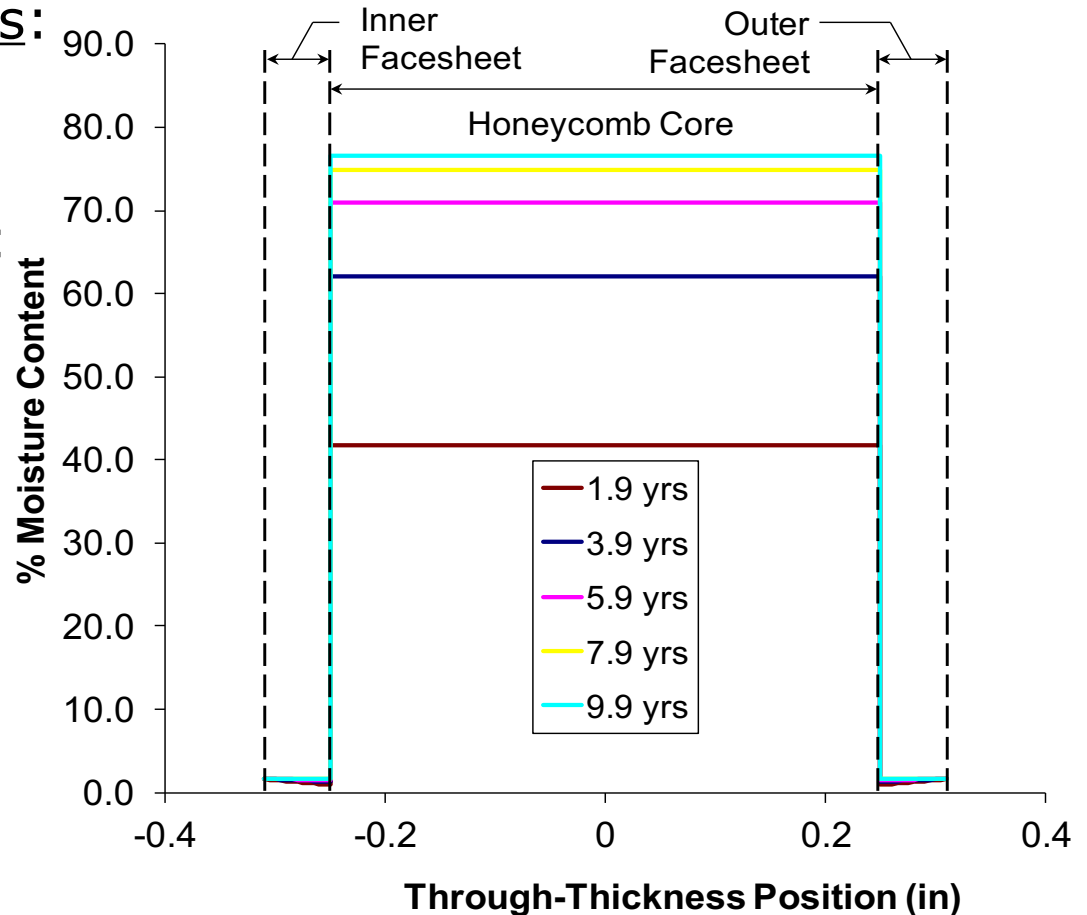


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Implication:

Since sandwich structures are exposed to low external temperatures during flight (-50°C), a condense-freeze-thaw-evaporate cycle will occur as internal humidity increases within the core volume of a sandwich composite, even if no mechanical damage of the facesheet has occurred

Objective of this study:

Determine if the condense-freeze-thaw-evaporate cycle is detrimental by monitoring room-temperature interfacial fracture toughness, G_c , associated with facesheet disbonding for four specimen types:

Type A: As produced, “dry” specimens

Type B: “Dry” specimens exposed to 300 thermal cycles from RT to -50°C

Type C: “Humid” specimens, with internal core humidity $\sim 80\%RH$

Type D: “Humid” specimens exposed to 700 thermal cycles from RT to -50°C

Effects of Moisture Diffusion in Sandwich Composites

- UW Principal Investigator and Researchers
 - Mark Tuttle (PI), Shuyu 'Frank' Xia (MSME), William Smoot (MSME), Sung Lin 'Jason' Tien (MSAA)
- FAA Technical Monitors
 - Lynn Pham, Zhi-Ming Chen
- Additional FAA Personnel Involved
 - Larry Ilcewicz, Curt Davies, Dave Westlund
- Industry Participation
 - Hamid Razi, Adam Sawicki, Bill Avery/The Boeing Company
 - Dan Holley and Chris Praggastis/3M
 - Bob Fagerlund/Bell Helicopter
- Study Initiated in September 2015

Effects of Moisture Diffusion in Sandwich Composites

Technical Approach

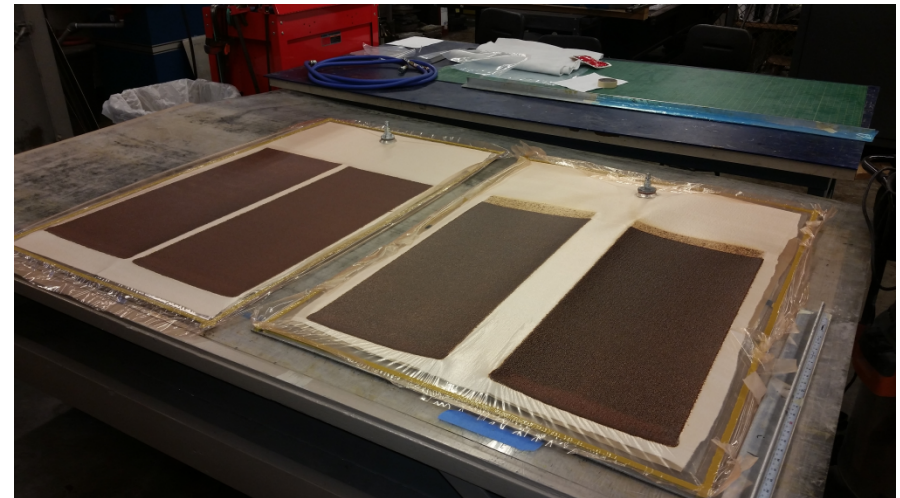
- Sandwich panels with 4-ply woven facesheets and $[45/0/0/45]_T$ stacking sequence:

Type	Manufacturer/Material Designation
Facesheet	Cytec T300/970 3k plain weave fabric
Core	Hexcel HRH-10 – 1/8 – 3.0 (0.50 in thick)
Adhesive	3M Scotch-Weld Structural Film AF 163-2K

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Technical Approach

- Facesheets produced using an autoclave:



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Technical Approach

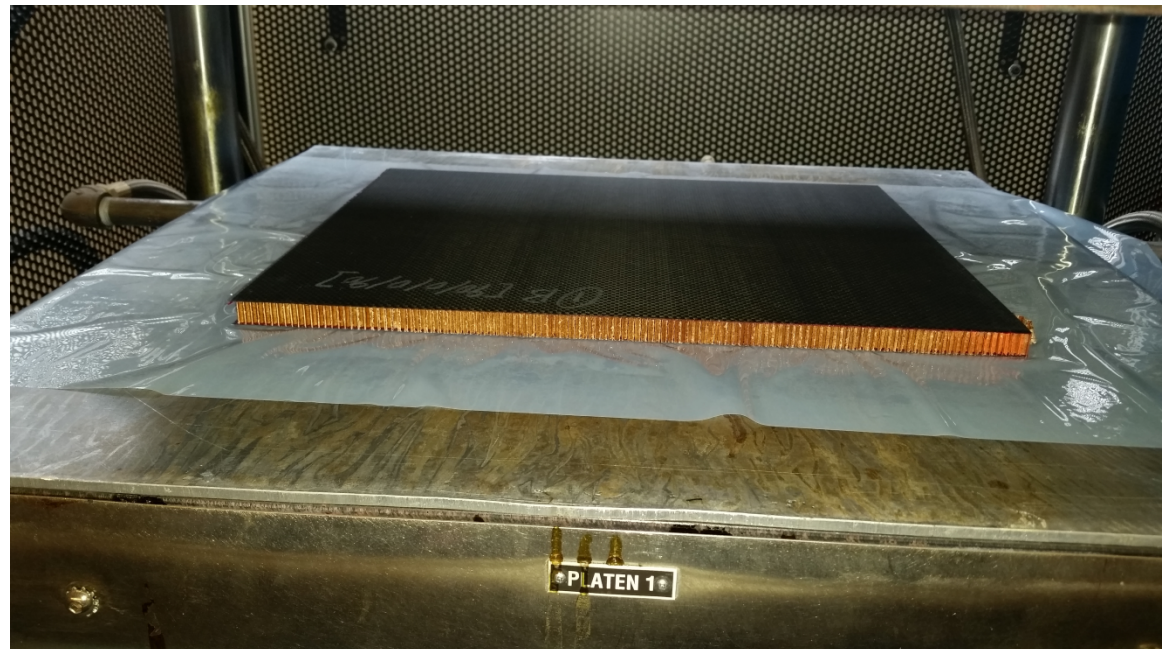
- The cured facesheets and Nomex core were machined to size and stored for 2 months at 50°C (122°F) at 8% RH in a humidity chamber, to insure components were as “dry” as possible



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Technical Approach

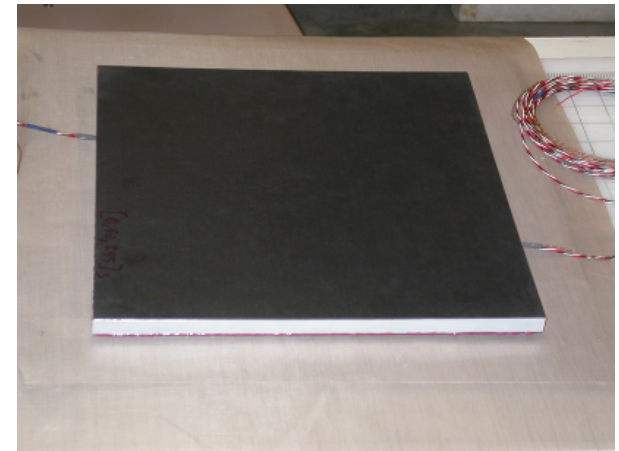
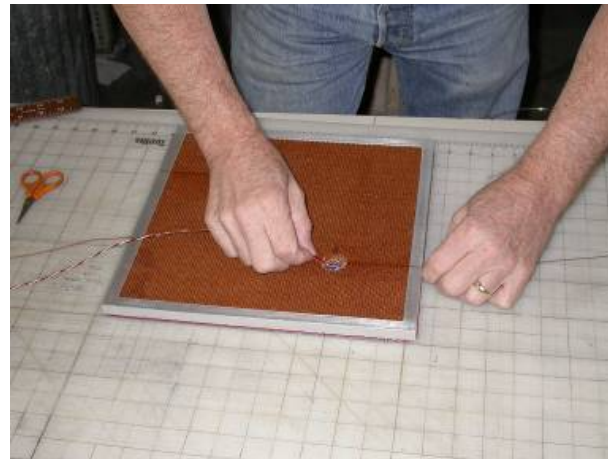
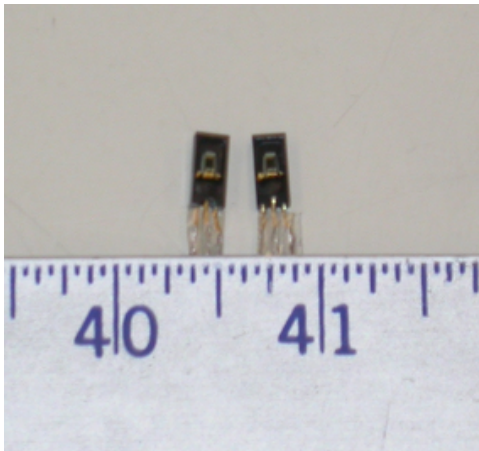
- Four parent sandwich panels were then produced using dried facesheets and core, using secondary bonding and a hot press



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Technical Approach

- Also fabricated instrumented “witness” panels with Ohmic Instruments Model HC-610 capacitive humidity sensors to monitor core humidity levels



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Technical Approach

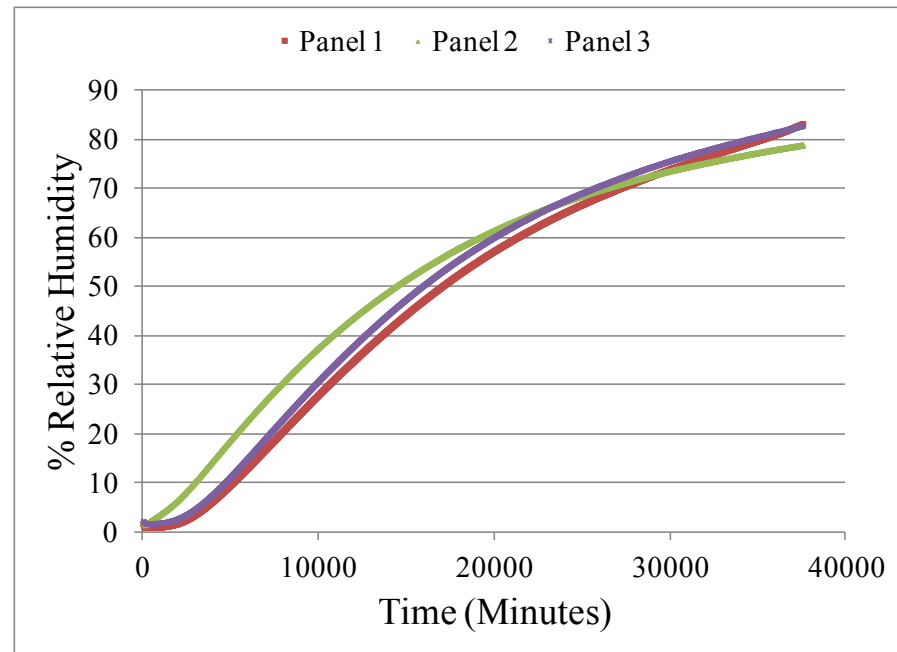
- Six tests specimens were machined from the four “parent” panels (24 test specimens in total)
- Specimens produced from each panel were used for each Type, to avoid any potential manufacturing bias

Type	Specimen Number					
A (as-produced)	1-1	2-2	3-3	4-4	1-5	2-6
B (thermally cycled)	2-1	3-2	4-3	1-4	2-5	3-6
C (humid)	3-1	4-2	1-3	2-4	3-5	4-6
D (humid&thermally Cycled)	4-1	1-2	2-3	3-4	4-5	1-6

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Technical Approach

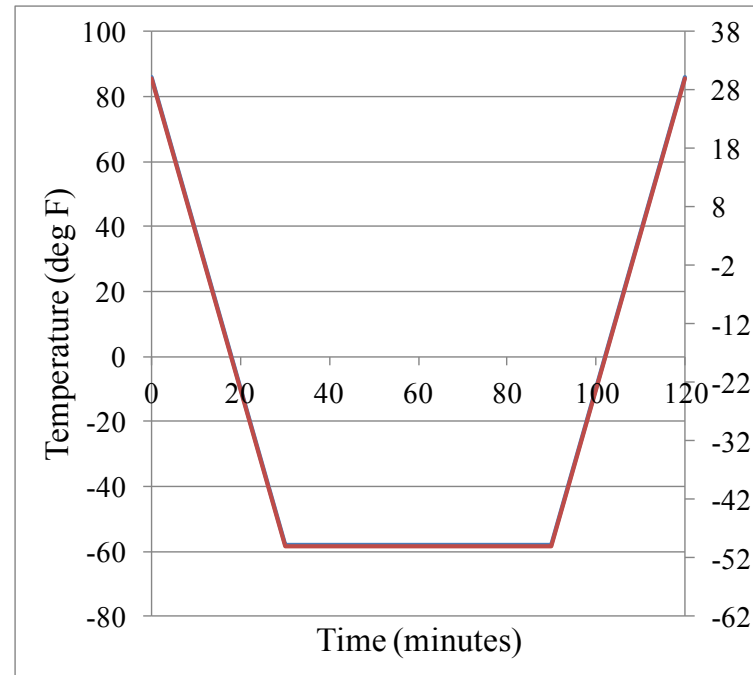
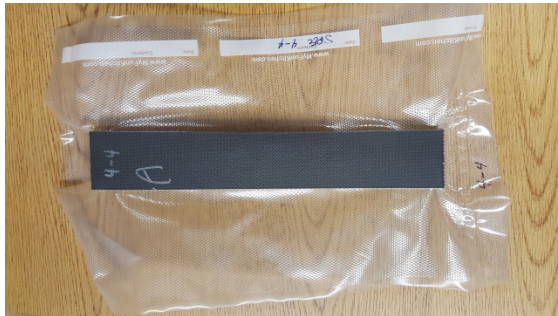
- Three witness panels and all Type C and D specimens were placed in the humidity chamber at 65°C (150°F) and 90%RH. Core humidity levels increased to about 80% in one month



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Technical Approach

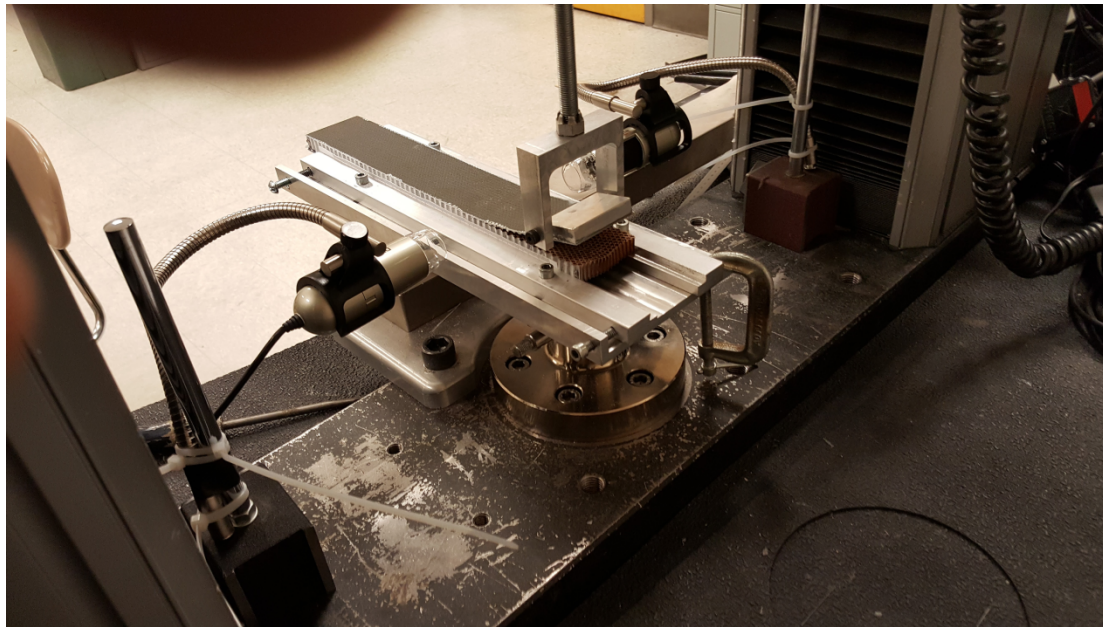
- All thermally-cycled specimens (Types B and D) were individually vacuum bagged (to insure constant moisture content in core volume) and subjected to 2-hr thermal cycles from 30°C ↔ -50°C



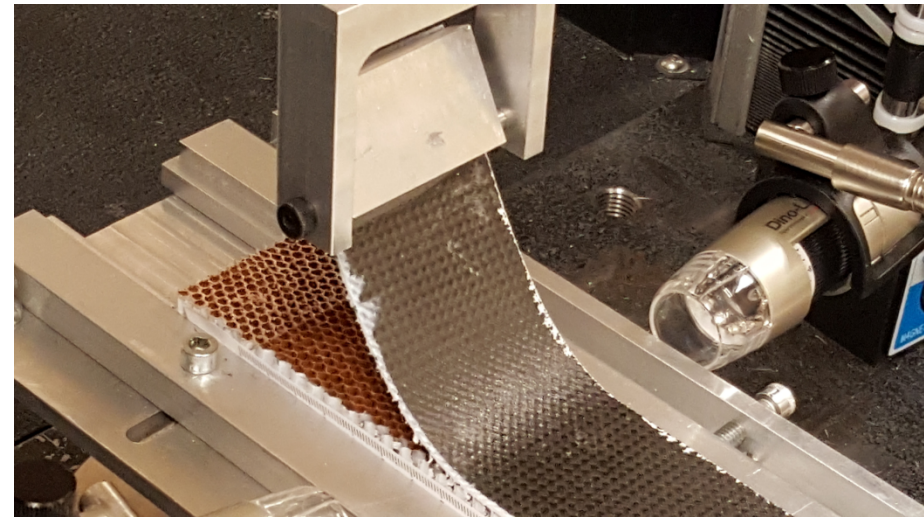
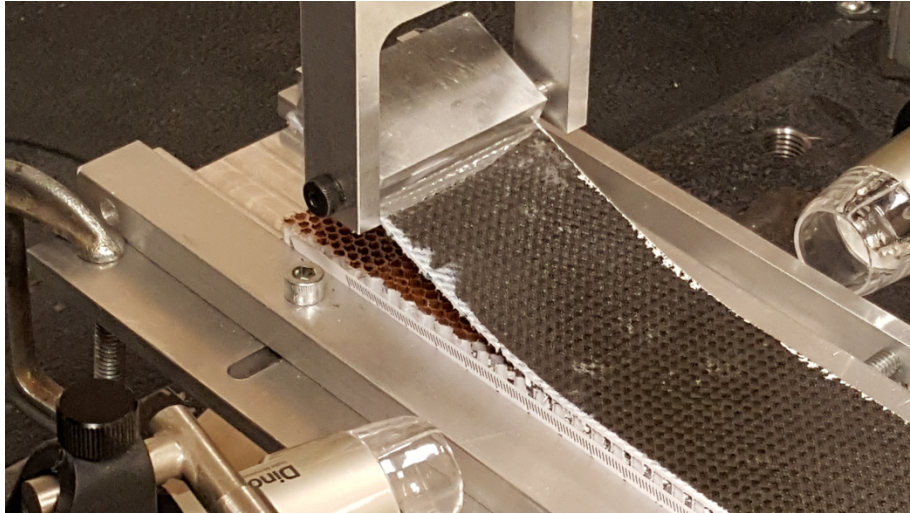
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Technical Approach

- The interfacial fracture toughness, G_c , was measured in accordance with a (draft) single-cantilever-beam (SCB) test standard being developed by Dan Adams, Waruna Seneviratne, and other members of a CMH-17 working group



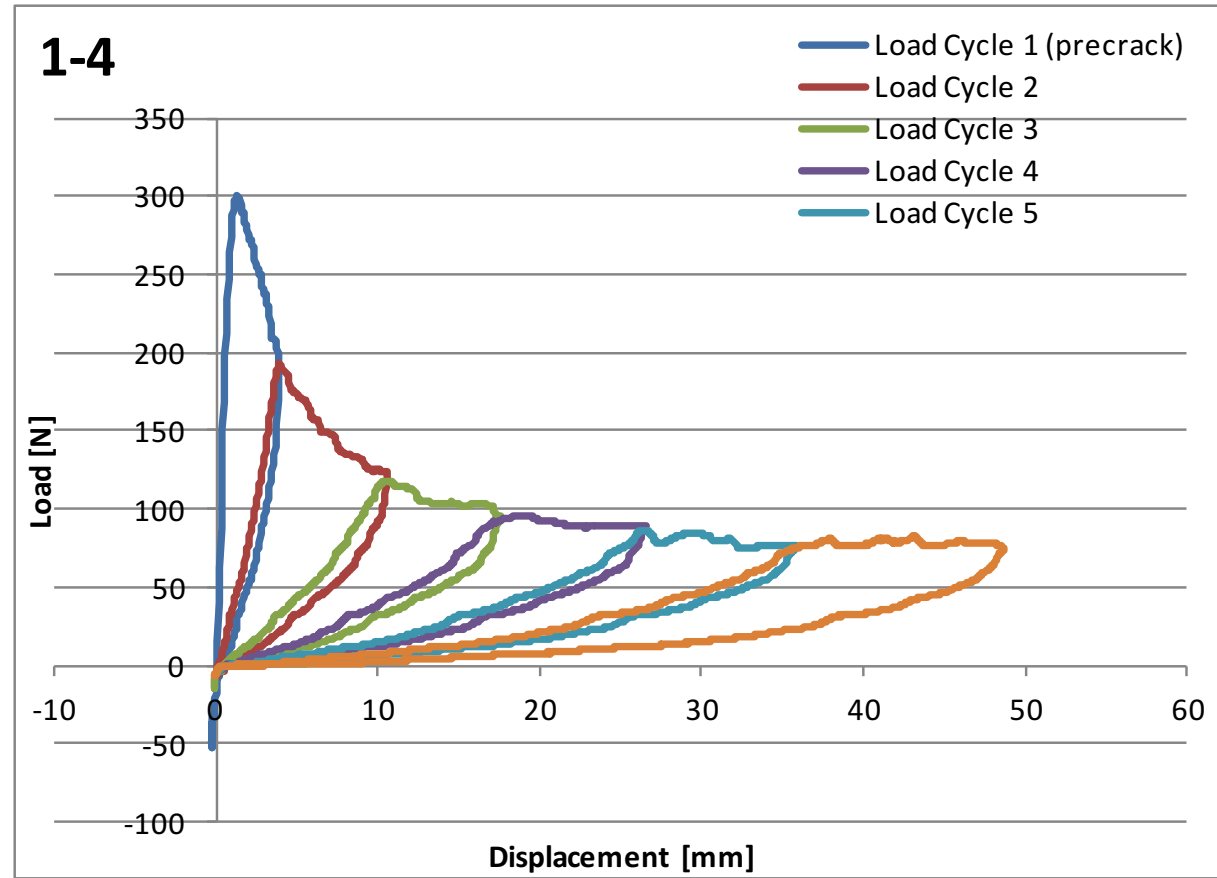
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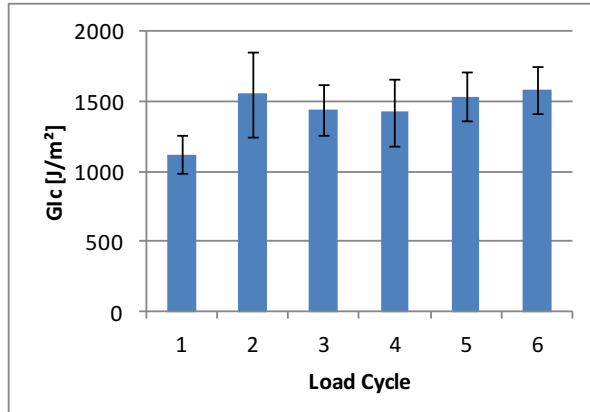
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Technical Approach

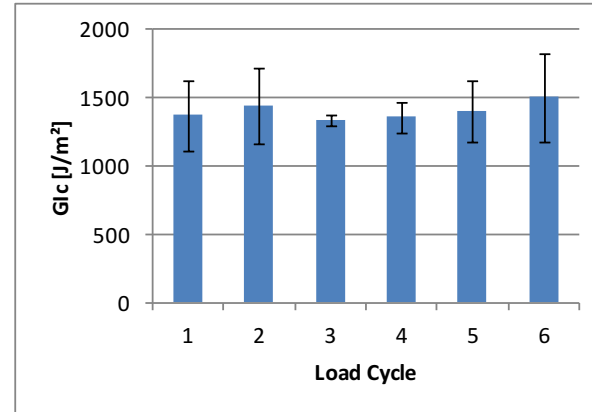
- A typical test involves six load cycles
- Crack length is measured after each cycle
- G_c can be calculated using data collected during any one of the six cycles (data from cycle 1 is normally discarded)



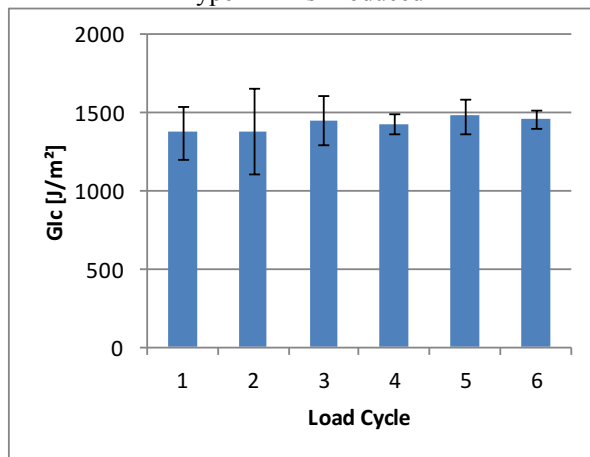
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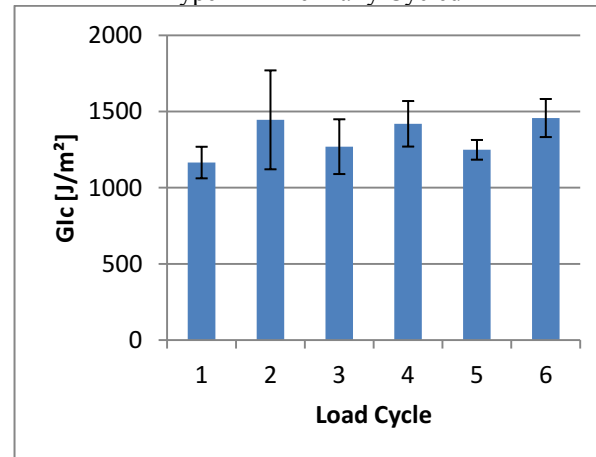
Type A - As Produced



Type B - Thermally Cycled

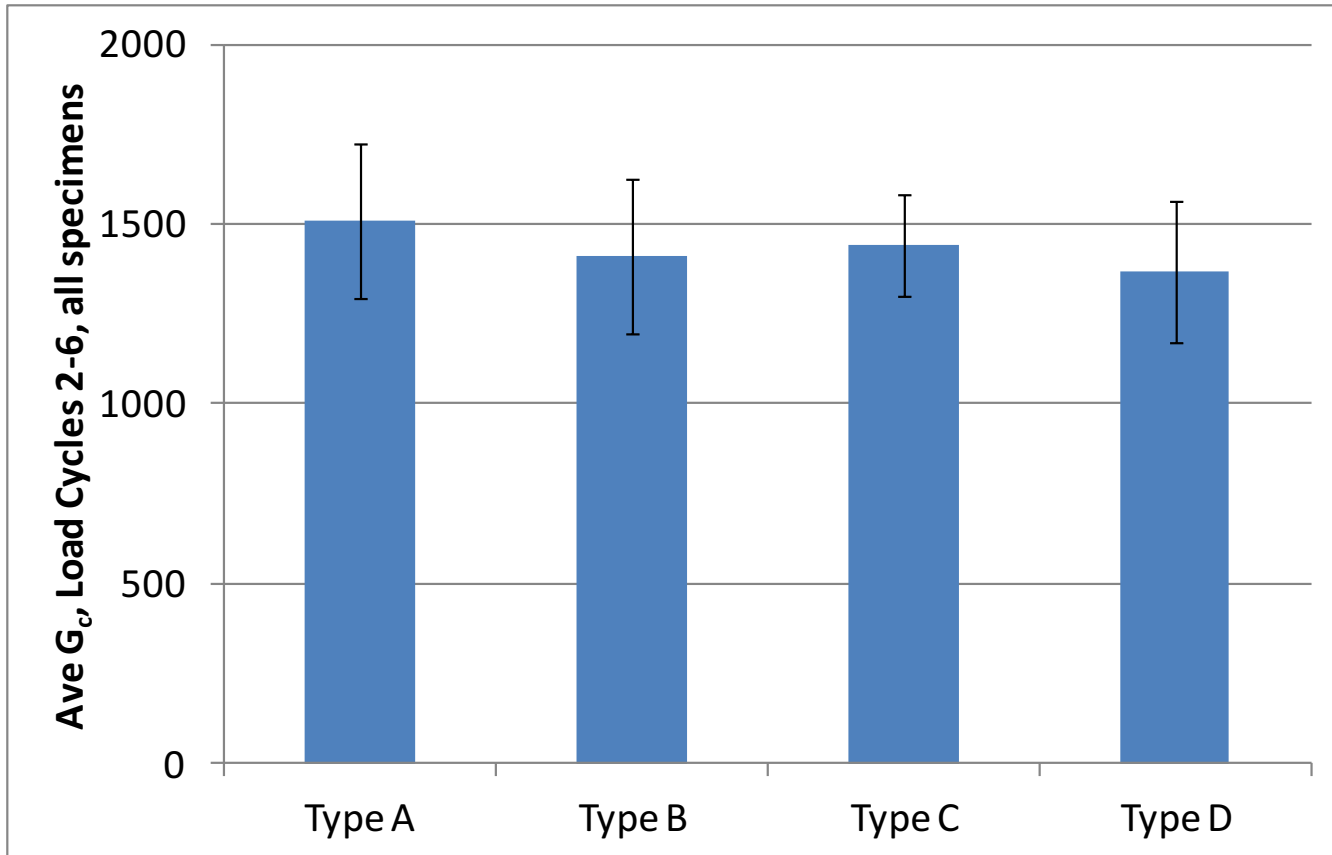


Type C - Humid



Type D - Humid and Thermally Cycled

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Effects of Moisture Diffusion in Sandwich Composites

Condition	Ave G_c (J/m ²)	StdDev G_c (J/m ²)	Average G_c , Normalized to Type A
Type A	1508	213	1.00
Type B	1410	214	0.94
Type C	1440	142	0.95
Type D	1368	198	0.91

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Conclusions

- Environmental factors (i.e., thermal cycling and/or elevated humidity levels) have a modest but measureable impact on interfacial fracture toughness, G_c , associated with facesheet disbonding in sandwich composites
- The most aggressive environmental conditions considered during this study (humid specimens exposed to 700 thermal cycles from RT to -50°C) resulted in about a 10% reduction in G_c .

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A Look Ahead

- Expand Test Matrix:
 - Kevlar honeycomb core
 - Different Nomex core densities (e.g., 8 lb/ft³)
 - Thinner (3-ply?) and thicker (8-ply?) facesheets
- Formally participate in CMH-17 sandwich disbond round-robin study

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Thank You!

Questions, Comments,
Suggestions?

Backup Slides

